



Runway Occupancy Time Estimation and SIMMOD

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Outline of this Presentation



- Review of past Runway Occupancy Time (ROT) research
 - Modeling approaches
 - Airport data collection
 - Flight simulation work
- Past ROT SIMMOD modeling activities at VPI
 - SIMMOD ROT pre-processor module development
 - SIMMOD engine runway module development
- Future steps

Definitions of ROT



- Landing ROT
 - Time spent on a runway from threshold crossing point until aircraft clears imaginary plane of the runway
 - NOTE: Our definition of ROT includes the turnoff time
- Departure ROT
 - Time spent on a runway from the clearance of imaginary runway plane until threshold crossing point (opposite end of the runway)
 -

Importance of Runway Occupancy Time



- Interarrival separation criteria dominates arrival service times (tendency to neglect ROT)
 - Some runways exhibit quite large variations in ROT
 - Large σ ROT (in seconds) is undesirable
- In mixed runway operations ROT matters as it affects the time to release departures (short ROT values are desirable)
- For VMC operations at large airports (say Atlanta) inter-arrival separations and ROT values are not very far (some go-arounds are possible in a day)

Typical ROT Values vs. Runway Exit Types



Scenario Number	Scenario	Runway Exits Description Exit Location (m.) Exit Type						Weighted Average ROT (s.)
		Exit # 1	Exit # 2	Exit # 3	Exit # 4	Exit # 5	Exit # 6	
1	Baseline	390 90 deg.	1154 90 deg.	1614 90deg.	2159 90 deg.	2713 90 deg.	3042 90 deg.	54.50
2	Wide Throat	390 90 deg.	950 WT	1225 WT	1425 WT	1900 WT	3042 90 deg.	51.20
3	30 Degree Standard FAA	390 30 deg.	950 30 deg.	1200 30 deg.	1400 30 deg.	1925 30 deg.	3042 90 deg.	44.63
4	30 Degree FAA Modified Exit ^a	390 90 deg.	900 30 deg. modified	1150 30 deg. modified	1350 30 deg. modified	1875 30 deg. modified	3042 90 deg.	43.00
5	REDIM 3030 ^b	390 90 deg.	875 RE 3020	1125 RE 3020	1325 RE 3020	1825 RE 3020	3042 90 deg.	40.80
6	REDIM 3530 ^c	390 30 deg.	825 RE 3520	1050 RE 3520	1250 RE 3520	1650 RE 3520	3024 90 deg.	36.80

a. The FAA modified 30 degree, acute angle geometry includes a 457 m. (1400 ft.) transition spiral.

b. The designation RE 3030 implies a high-speed exit designed for 30 m/s entry speed and a 30 degree exit angle.

c. The designation RE 3530 implies a high-speed exit designed for 35 m/s entry speed and a 30 degree exit angle.

Previous Research in ROT at VPI



- Development of the Runway Exit Design Interactive Model (REDIM) - 1990-1993
 - Report site: <http://www.ce.vt.edu/faa/redim.html>
- Calibration of REDIM with field observations - 1993-1994
- Investigation of high-speed runway exits using an FAA Boeing 727 aircraft simulator (6-DOF simulator) - 1995
 - Report site: <http://www.ce.vt.edu/faa/fltsim.html>
- Implementation of SIMMOD engine changes (at VPI) using UNIX MOTIF GUI standards - 1996
 - Report site: <http://www.ce.vt.edu/faa/simmod.html>

REDIM Model



- Provides an estimation of ROT values for a single runway and a user-defined aircraft mix (FAA/NASA sponsors)
- Provides guidance on optimal runway exit locations

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UPI&SU <<<< RUNWAY EXIT DESIGN INTERACTIVE MODEL >>>> UCTR
WELCOME TO REDIM VERSION 2.0 !!!

FAA NASA VT

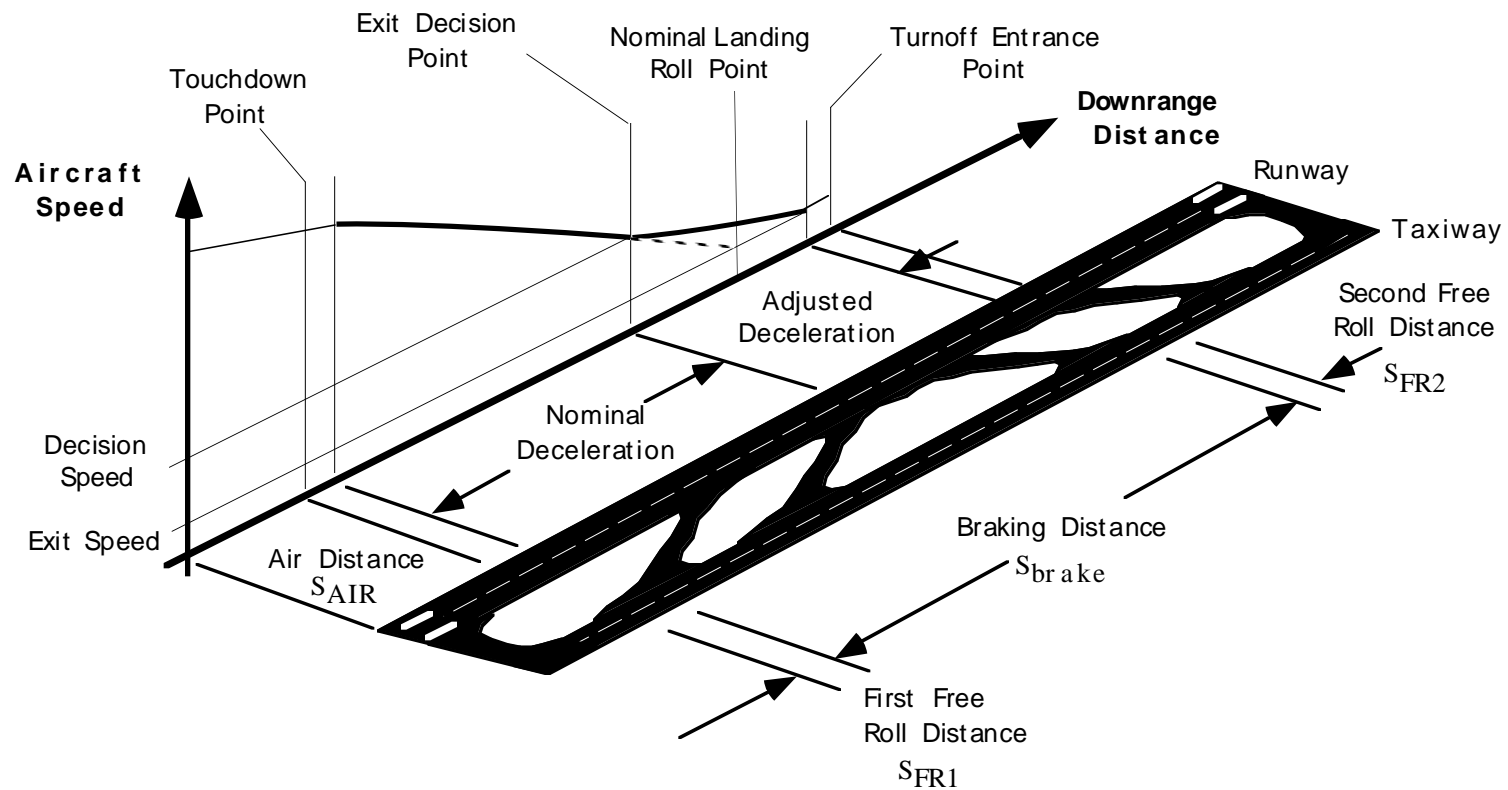
AVAILABLE DATA FILES IN THIS DIRECTORY
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TEST1   DCA36   KIM

Enter the file name you desire to work -->■
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Aircraft Landing Roll Analysis



- Characterization of aircraft landing roll profiles



Aircraft Mix Selection Menu



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UPI&SU <<<< RUNWAY EXIT DESIGN INTERACTIVE MODEL >>>> UCTR
INPUT the percentages of aircraft: Press <Tab> key after editing.
Peak Period Operation Mix

TERPS-A          TERPS-B          TERPS-C          TERPS-D          TERPS-E
BE-F33A          : 10             BE-58             : 0               Avro_RJ85        : 0               A330-300         : 0               F-4              : 0
CE-172           : 0               BE-300           : 0.7             A300-600        : 0               A340-200         : 0
CE-182           : 0               BE-400           : 0.4             A310-300        : 2.6             B-747-200        : 0
CE-208           : 0               CE-402           : 0.7             A320-200        : 0.7             B-747-400B      : 0
CE-210P         : 0               CE-421           : 0               B717-200        : 0               B767-300         : 0
PA-288-161      : 0               CE-550           : 0.4             B727-200        : 0               B777-200         : 0
PA-288-236      : 0               CE-650           : 0               B737-200        : 28.4            C-5A             : 0
PA-32-301       : 0               CRJ-200          : 0               B737-300        : 0               DC-8-73          : 0
PA-38-112       : 0.4            DA-200          : 0               B737-800        : 4               DC-10-30         : 0
PA-46-310P     : 0               DHC-7           : 0.7             B757-200        : 9.5             MD-11            : 0
DHC-8           : 0               DHC-8           : 0               Bae-146-200    : 0               L1011-500       : 0
Do-328         : 0               Do-328         : 0               CL-601-3A       : 20.3
Do-728         : 0.4            Do-728         : 0.4            DC-9-32         : 0
EMB-120        : 0               EMB145         : 1.9             Fokker-100      : 12.2
Lear-31         : 0               Lear-31         : 0               Gulfstream       : 0
PA42-1000      : 1               Saab-340       : 2.6             IAI-1124        : 0.4
SA227          : 0               Shorts-330     : 0.4            Lear55          : 0
MD-83          : 0               MD-87          : 0
MD-87          : 0

```

NO. of Aircraft = 20
TOTAL Percentage = 97.7

Typical REDIM Results



- Sample ROT table for various aircraft

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UPI&SU <<<< RUNWAY EXIT DESIGN INTERACTIVE MODEL >>>> UCTI
Edit      Analysis  Output   Print    Newfile  Quit
    
```

ROT/ASSIGNMENT TABLE
(This is for Evaluating an Existing Runway)

Exit # Location (m)	1 600.0	2 940.0	3 1008.0	4 1368.0	5 1507.0	6 2078.0
Open/Close Exit Type	Close U-Def	Open 90-Deg	Open U-Def	Open U-Def	Open U-Def	Open 90-Deg
B737-800						
DRY ROT (4.1%)				47.78 96.0%	51.15 4.0%	
WET ROT (0.0%)				47.99 62.0%	51.92 35.0%	67.93 3.0%
B757-200						
DRY ROT (9.7%)				48.23 100.0%		
WET ROT (0.0%)				48.31 86.0%	52.02 14.0%	
CL-601-3A						
DRY ROT (20.8%)		36.42 100.0%				
WET ROT (0.0%)		36.92 85.0%	36.22 15.0%			

ROT - Runway Occupancy Time in Secs Weighted Average ROT = 45.72

ROT Airfield Calibration Study

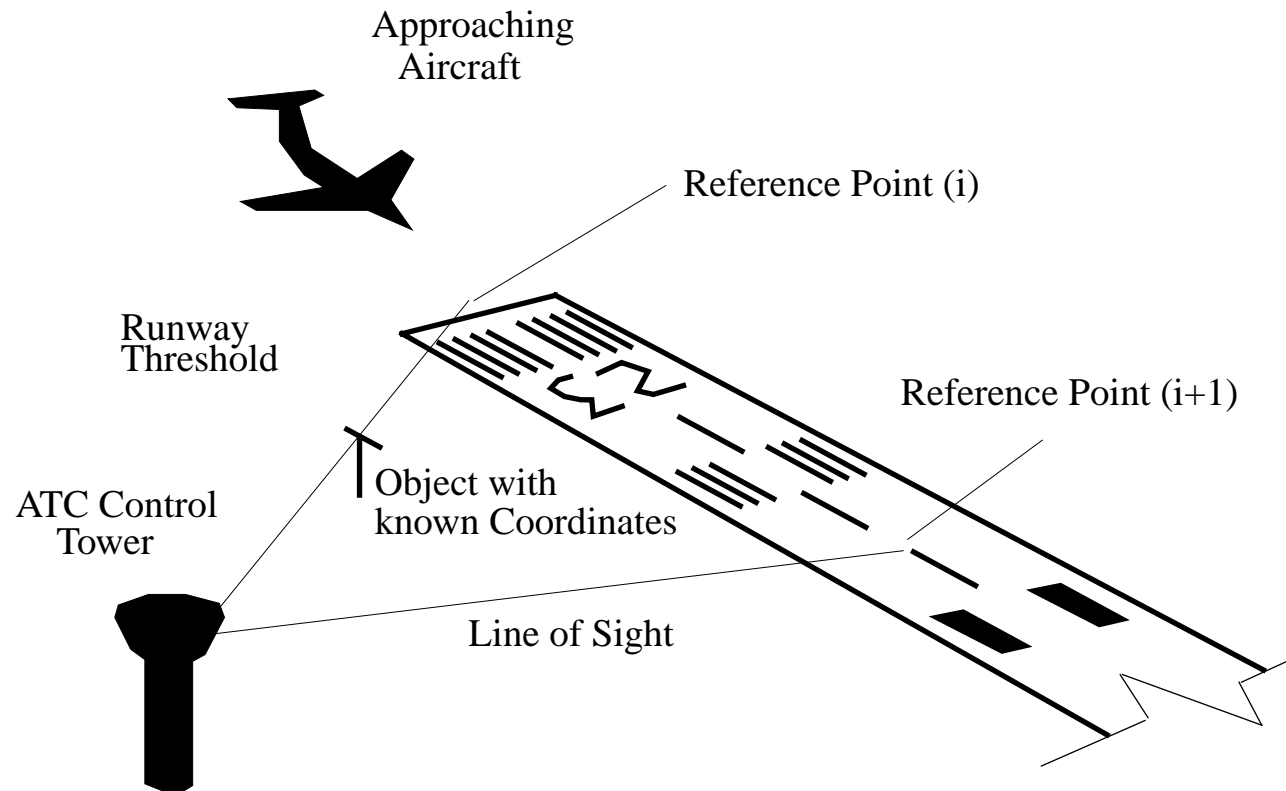


- Collected data at five airports in the U.S. East Coast
 - RDU, DCA, IAD, CLT and ATL
 - 2,400 landings under dry and wet pavement conditions
 - Use of video equipment to study aircraft time-space motion
- Developed a simplified ROT runway model (that can be programmed in a spreadsheet or numerical package see page 18 of this presentation)
- Incorporated heuristic rules into REDIM

Data Collection Method



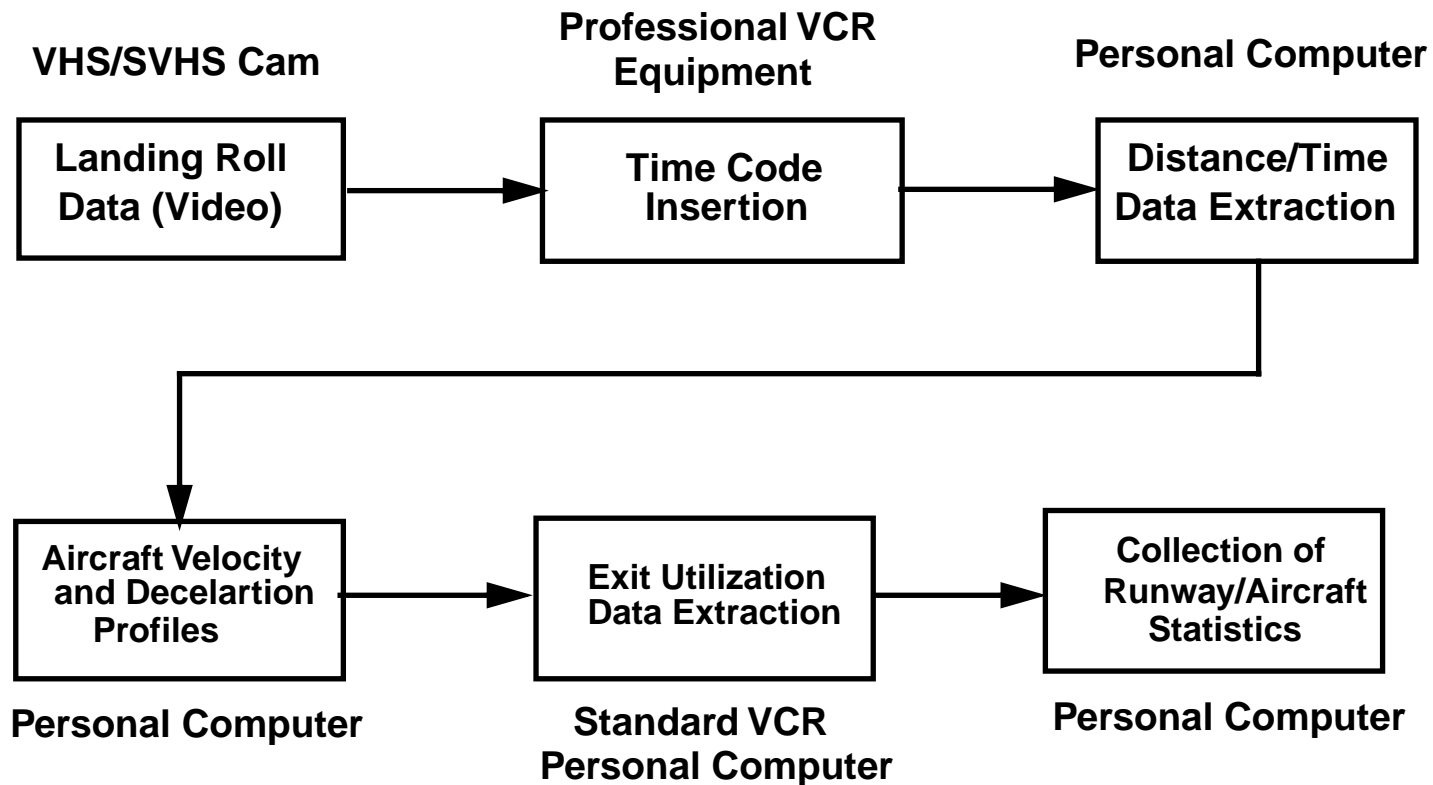
Use of video cameras located at the ATC control tower to assess ROT with precision



Data Analysis Techniques



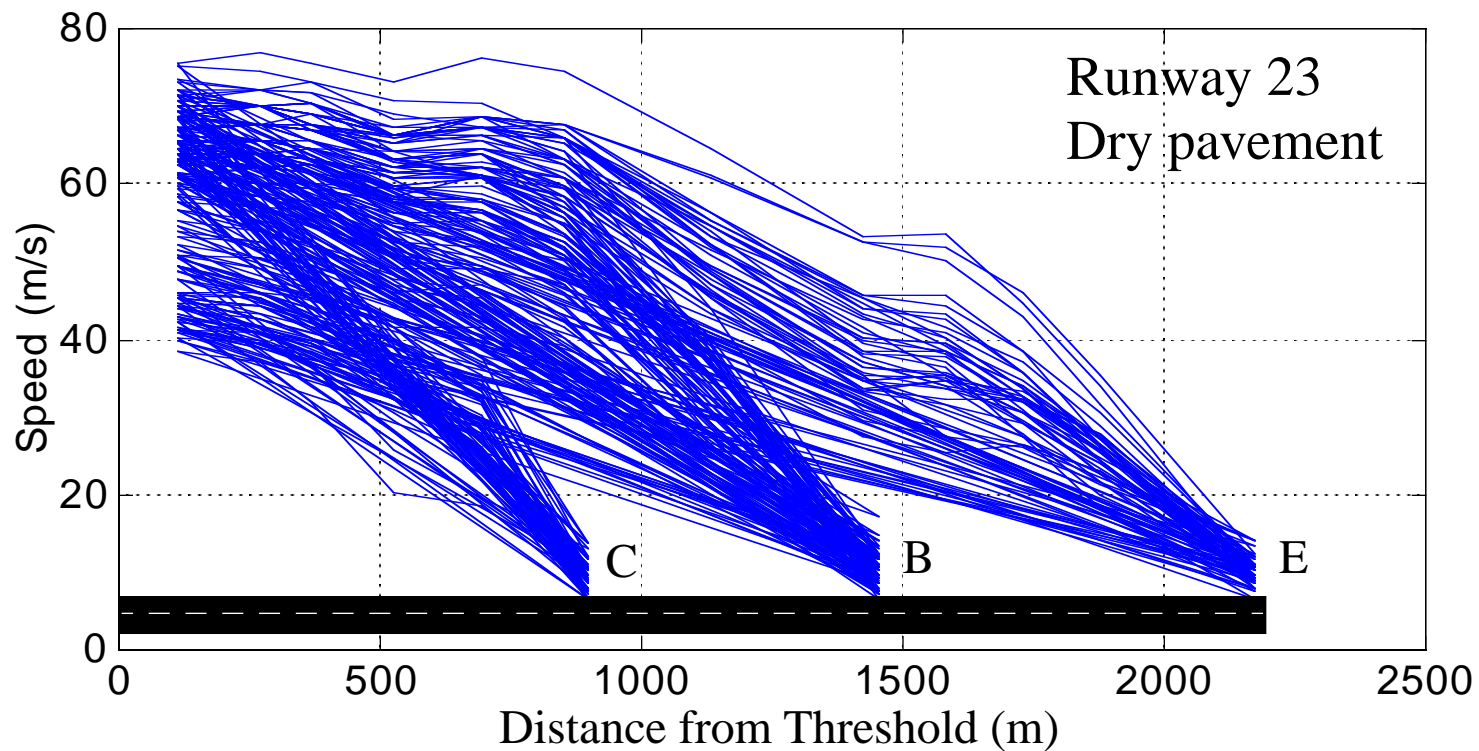
Use of computers and video editing hardware to extract ROT



Sample Runway Velocity Profiles



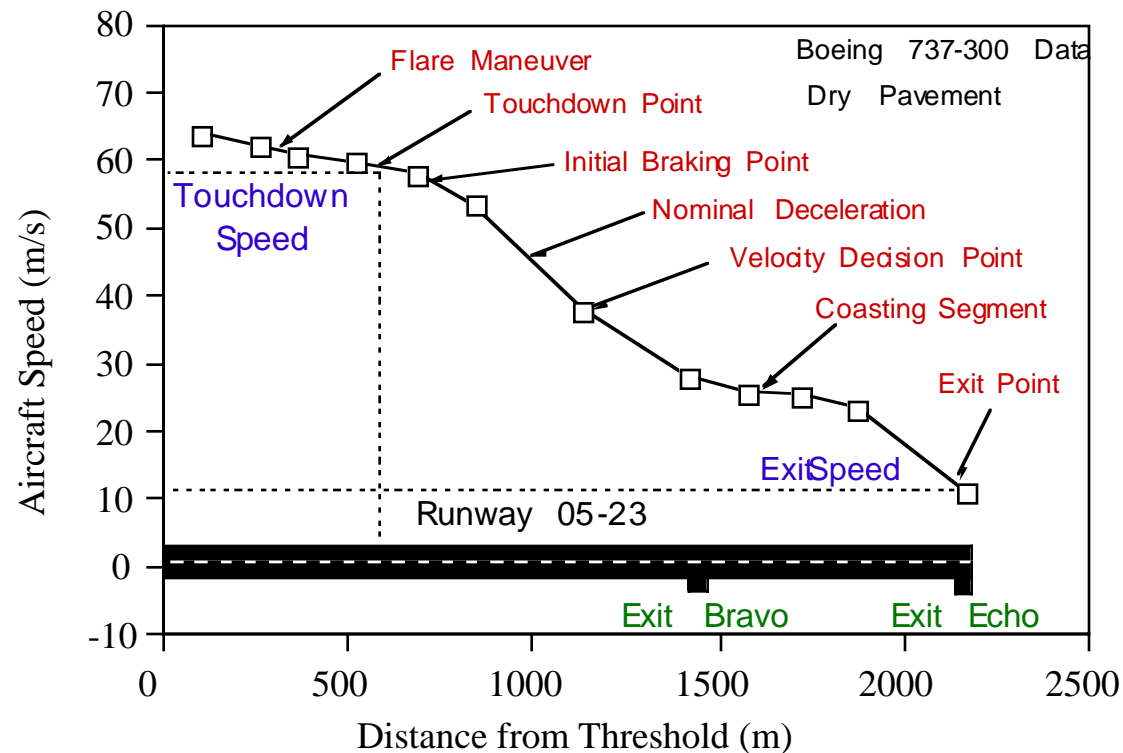
- Aircraft display large differences in landing roll behavior at real airports (shown data for CLT)



Aircraft Velocity Profile Phases



- A sample distance-velocity profile showing all aircraft landing phases modeled



Statistical Analysis



- Parameter statistical correlation (2-way ANOVA)

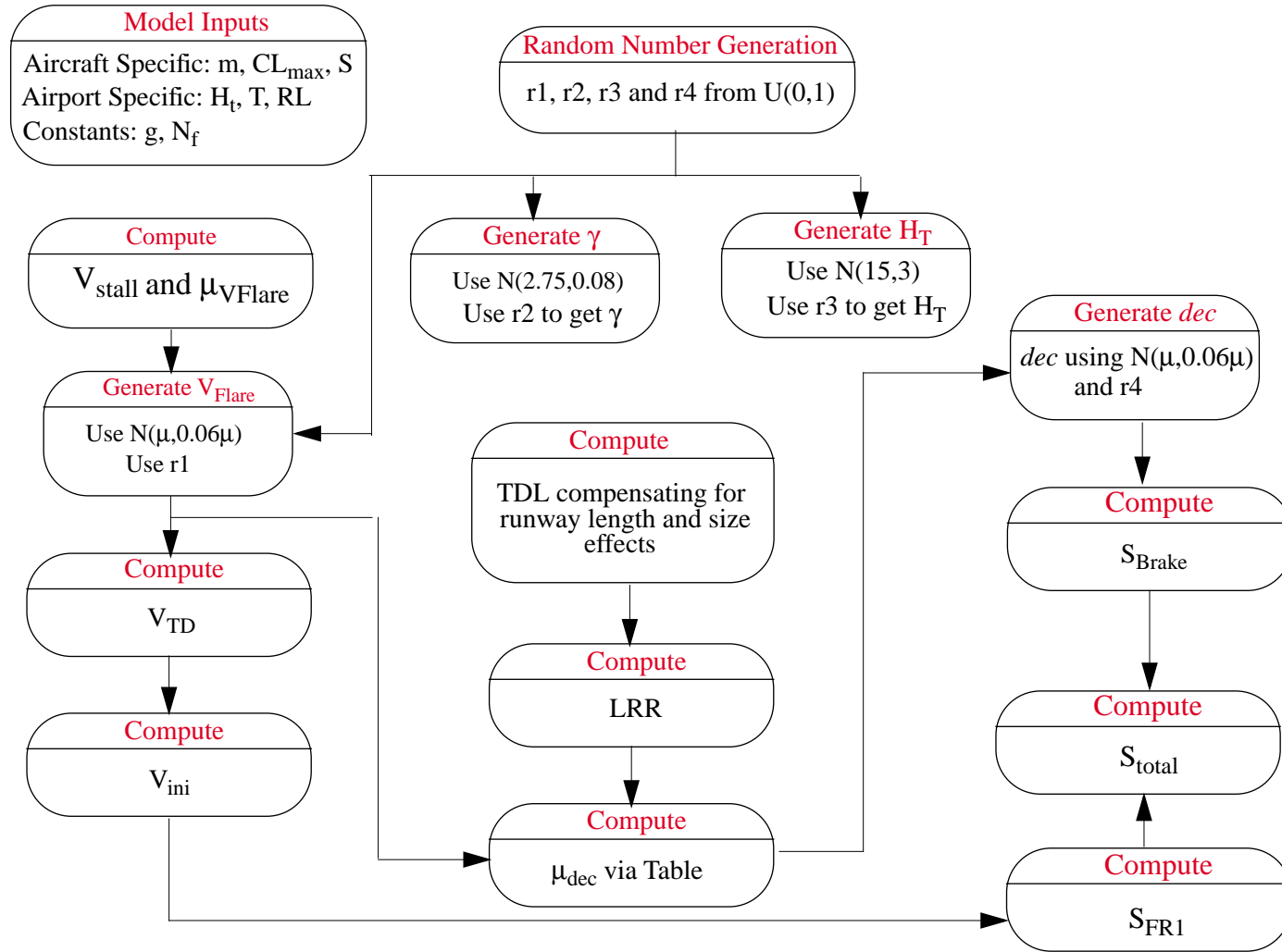
Variable	Factor	F-statistics	F-criteria ^a	p-value	Conclusion ^b
Flare Speed (m/s)	A/C Type	5.44	2.37	0.0003	H ₁
	Airport	8.59	3.00	0.0002	H ₁
	Interaction	2.00	1.94	0.0461	Marginal
T. D. Location (m)	A/C Type	3.95	2.37	0.0038	H ₁
	Airport	69.09	3.00	0.0001	H ₁
	Interaction	2.36	1.94	0.0173	H ₁
Braking Dec. (m/s ²)	A/C Type	3.64	2.37	0.0064	H ₁
	Airport	11.36	3.00	0.0001	H ₁
	Interaction	1.10	1.94	0.3617	H ₀

a.F-criteria is set for 5% significance level

b.H₀: the factor does not affect the variable in question (H₁: not H₀)



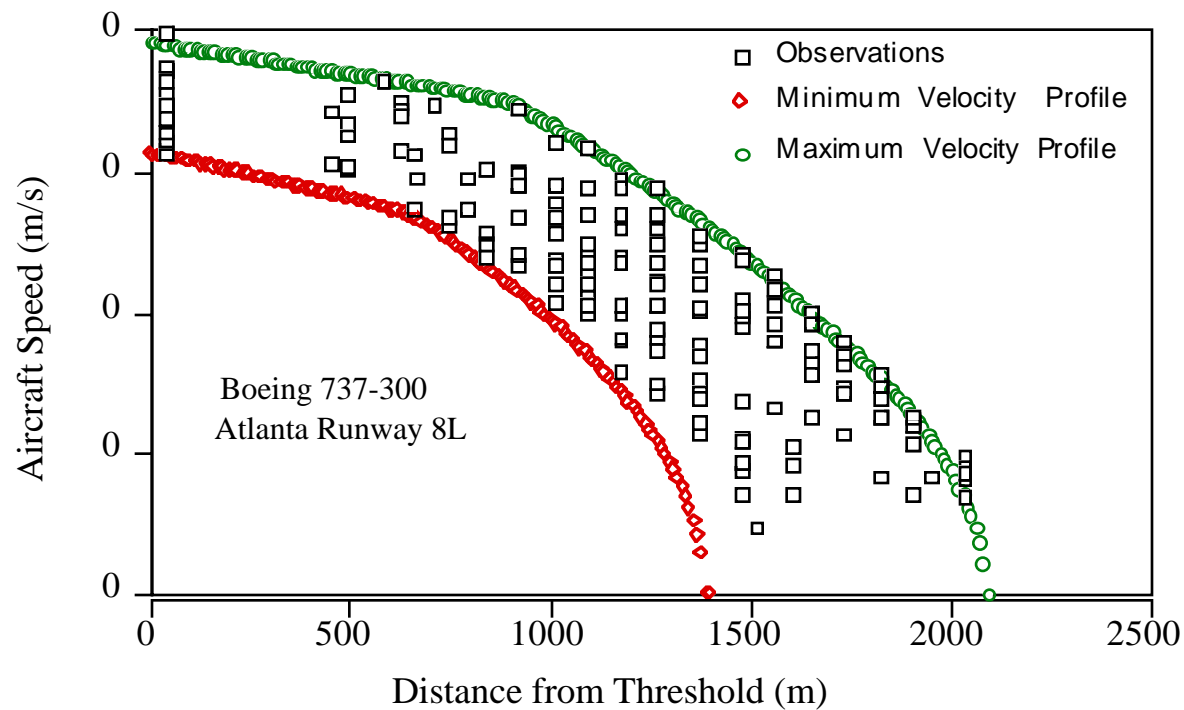
Simple Model Computational Procedure



Mathematical Model vs. Observations



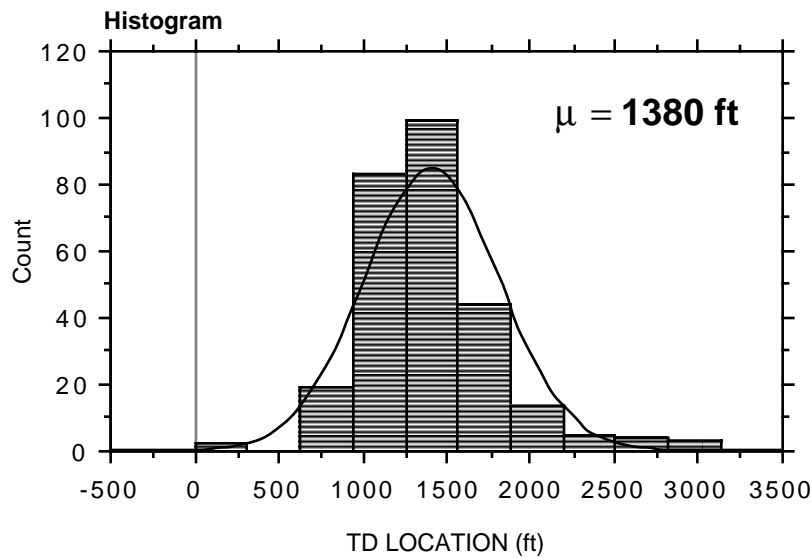
- Model comparison with actual aircraft landing roll observations (Boeing 737-300 landings at Atlanta 8L)



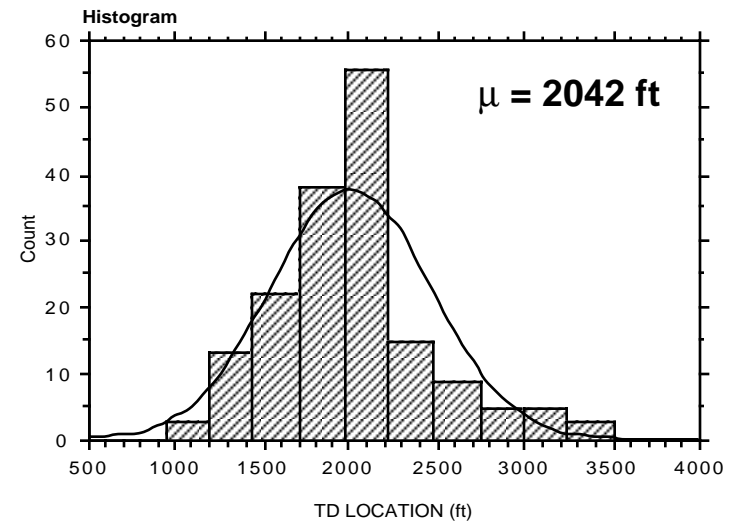
Sample Results of the Runway Module



Model adaptation to various runway length conditions

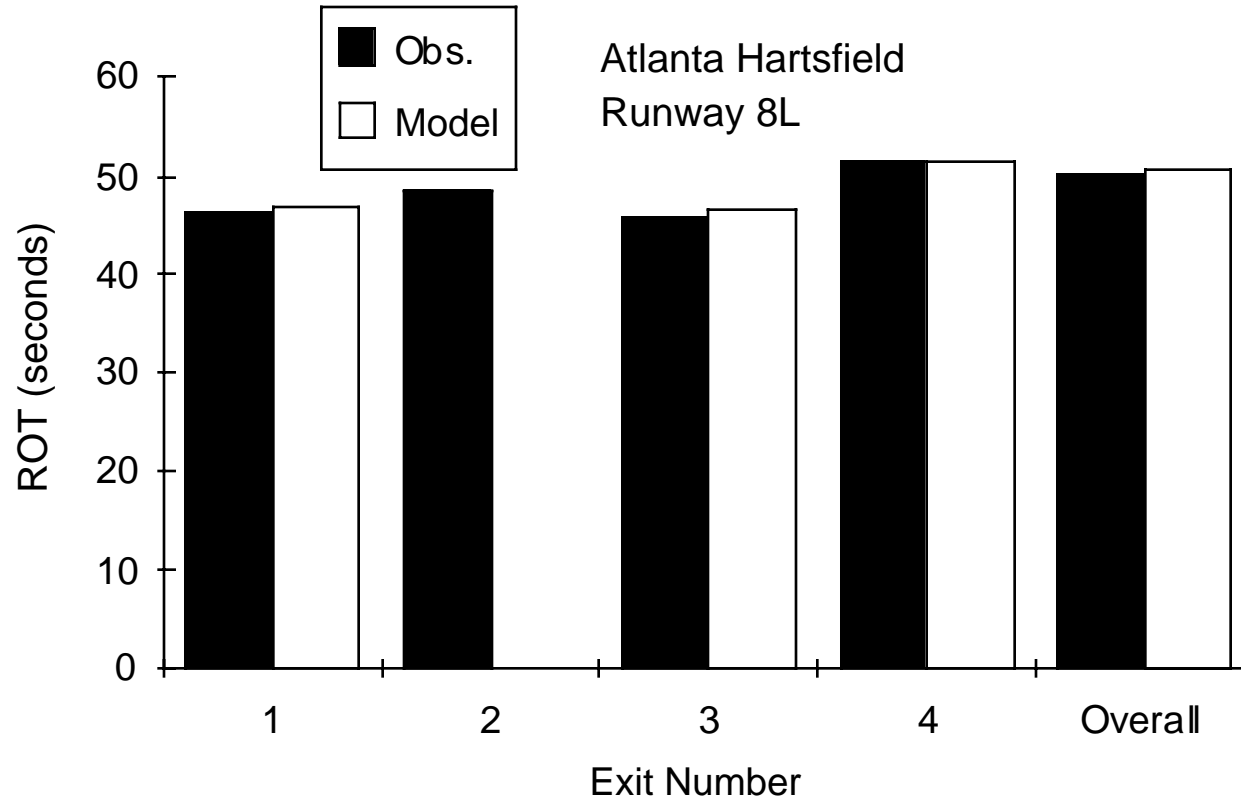


National Runway 36 (6700 ft)

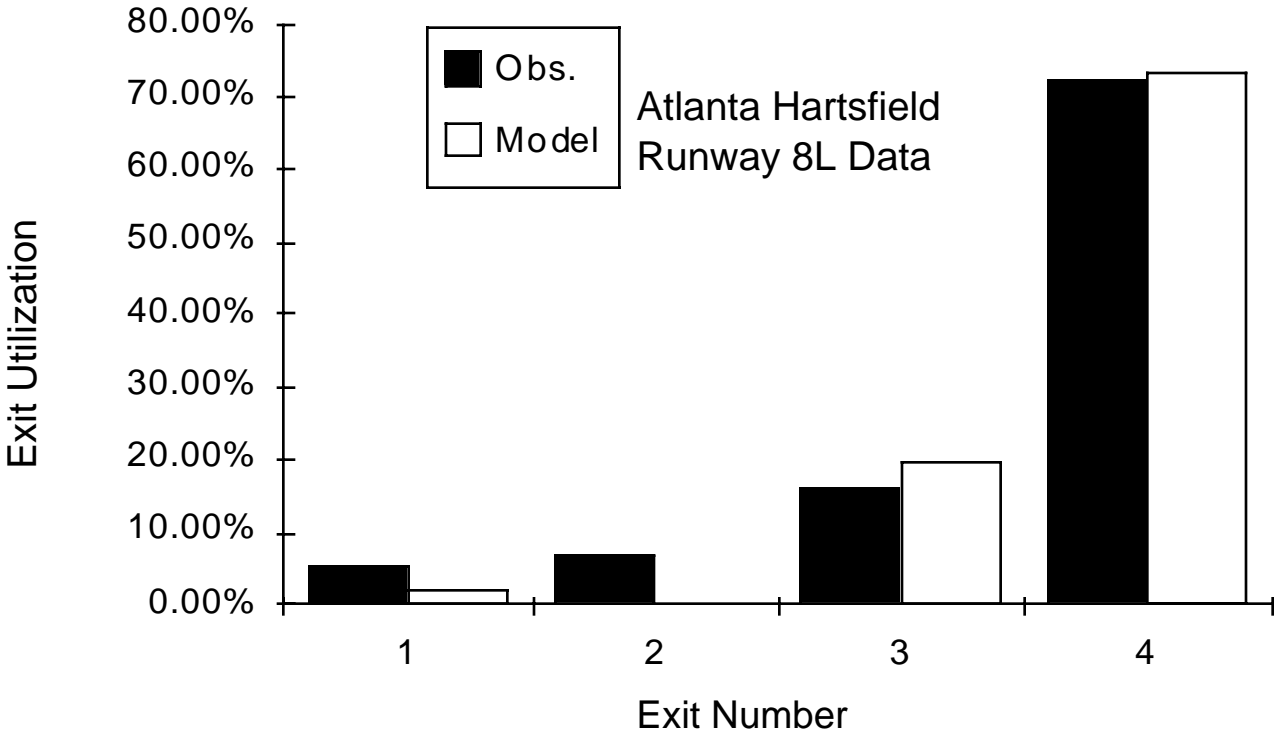


Atlanta Runway 8L (10,000 ft runway)

Runway Occupancy Time Predictive Capabilities



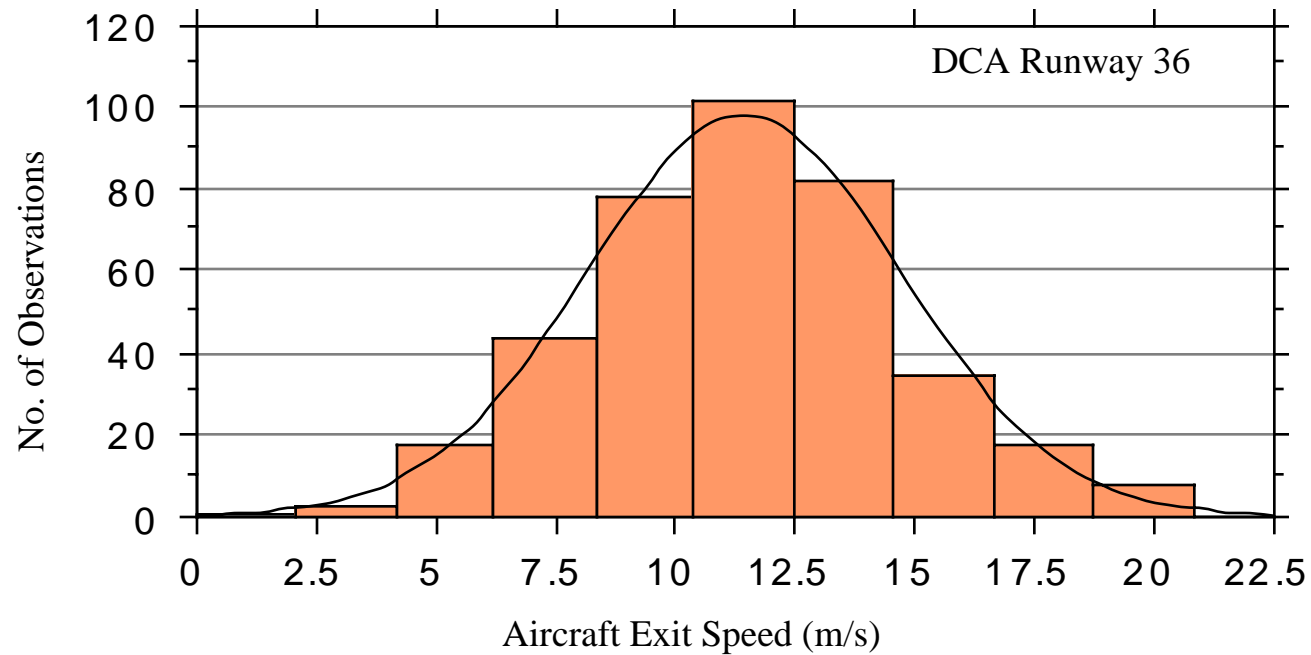
Runway Exit Utilization Results



Runway Exit Data Collection



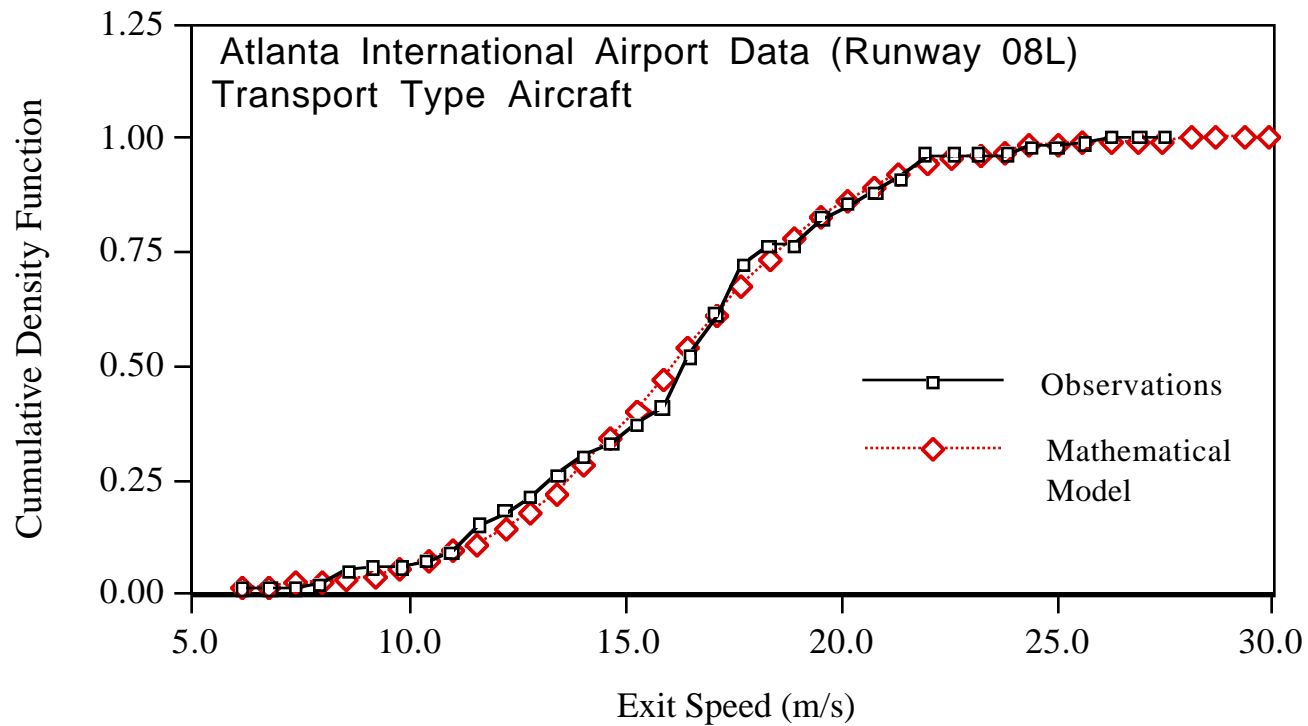
- Runway exit data has also been modeled mathematically



Runway Exit Mathematical Modeling



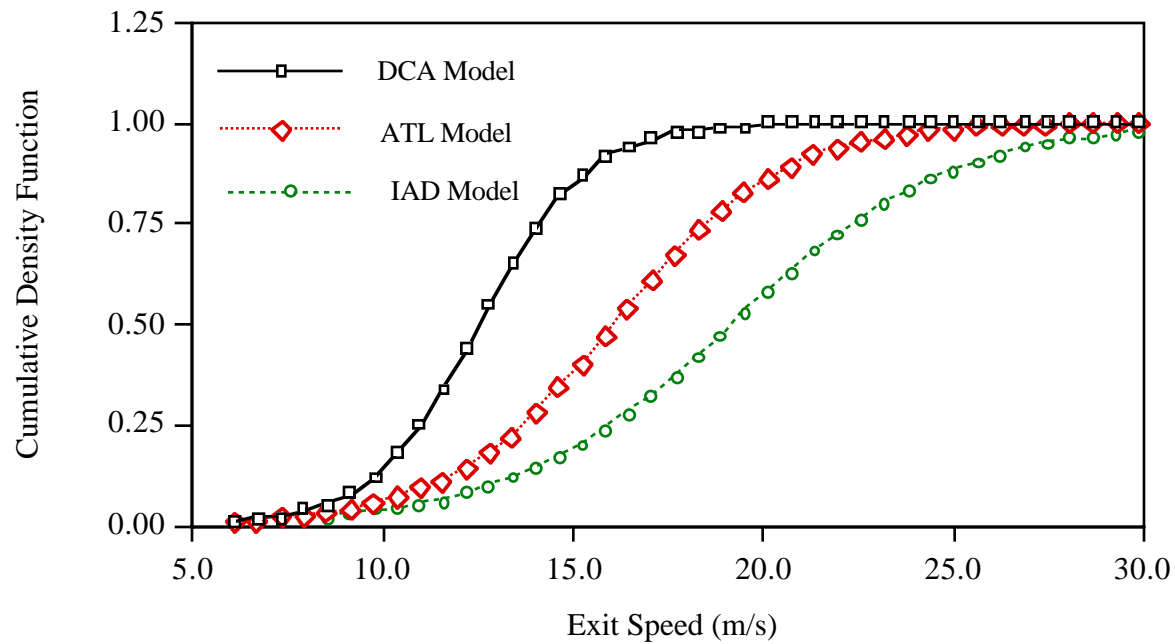
- In many instances the runway exit data was observed to follow normal distribution trends



Runway Exit Speed Models



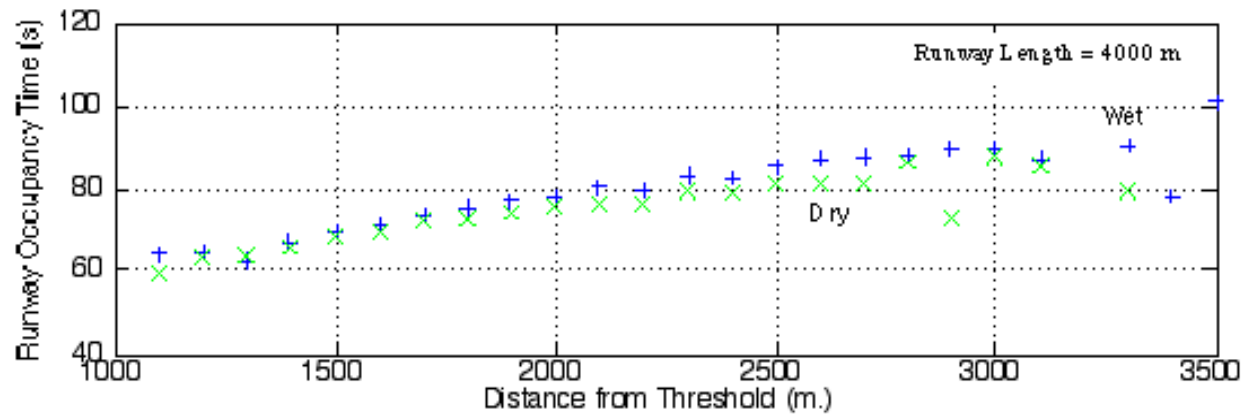
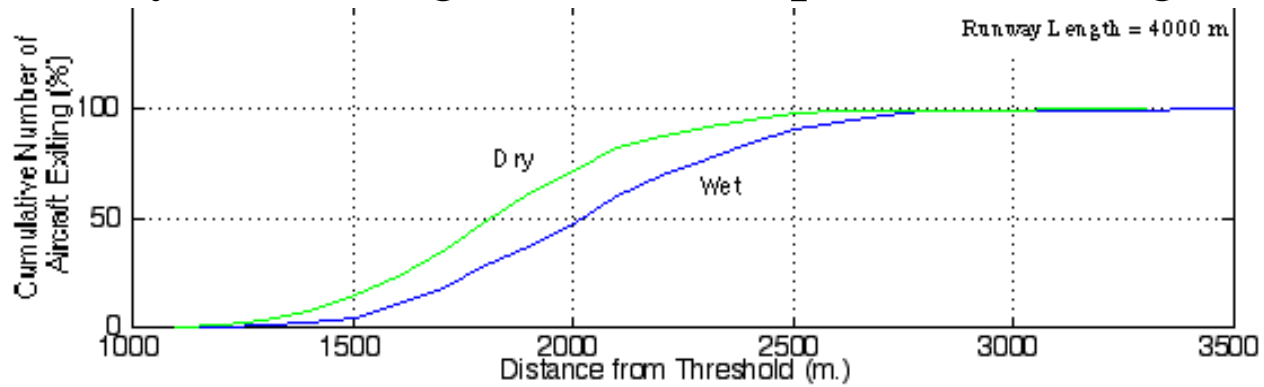
As expected, runway exit speeds vary drastically with the geometric features of the turnoff and the available lateral distances





ROT and Exit Assignment Plots

- Runway Exit Assignment (Group D and 90 deg. turnoff)



Conclusions from Airfield ROT Data



- Individual aircraft show dissimilar landing roll characteristics at various airports
 - flare speed
 - touchdown location
 - runway deceleration, etc.

The individual aircraft modeling scheme adopted in REDIM and later in the SIMMOD engine seems justified

SIMMOD Runway Module Development



Sponsored by FAA (Steve Bradford - technical monitor)

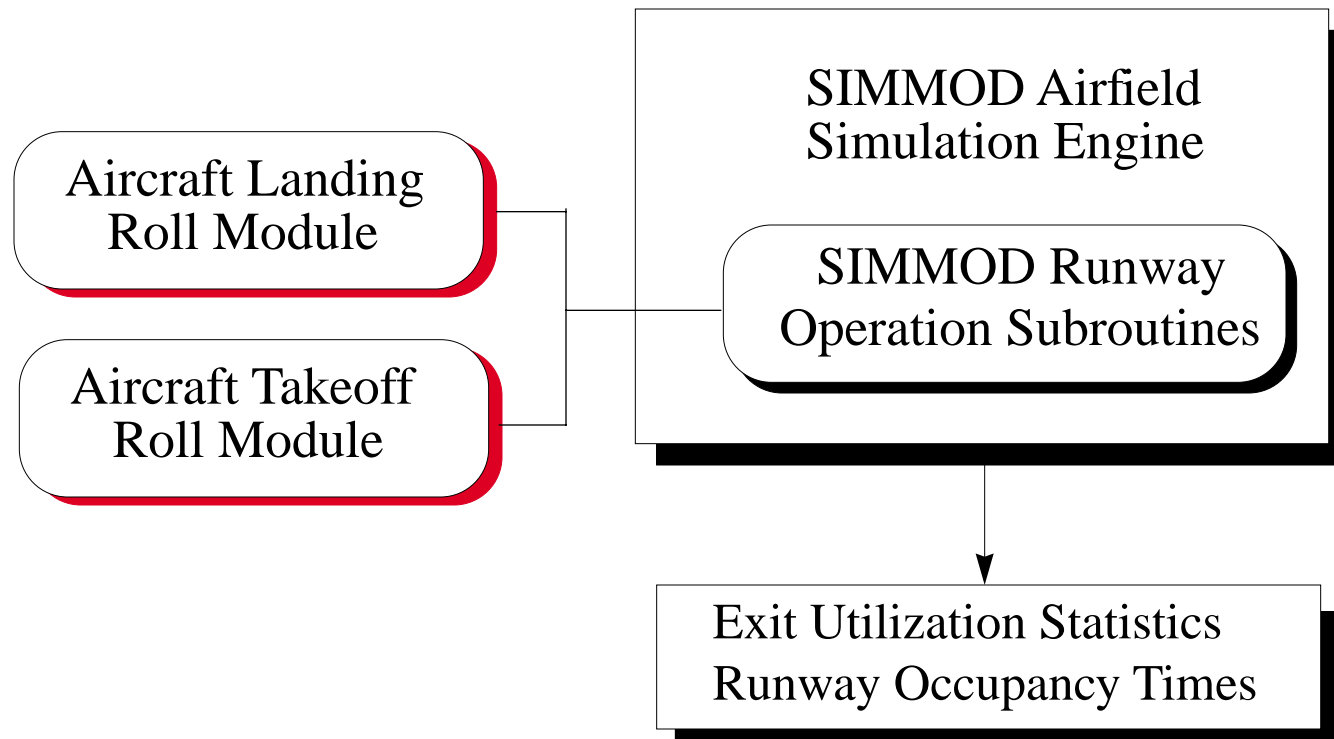
Main goals of this project:

- Development a more realistic runway operations model (based in algorithms developed for REDIM) and its incorporating into the SIMMOD engine
- Develop a pre-processor to execute all tasks available in REDIM (i.e., runway evaluation, runway enhancement and optimization of ROT parameters)

Model Applications (Runway Operations)



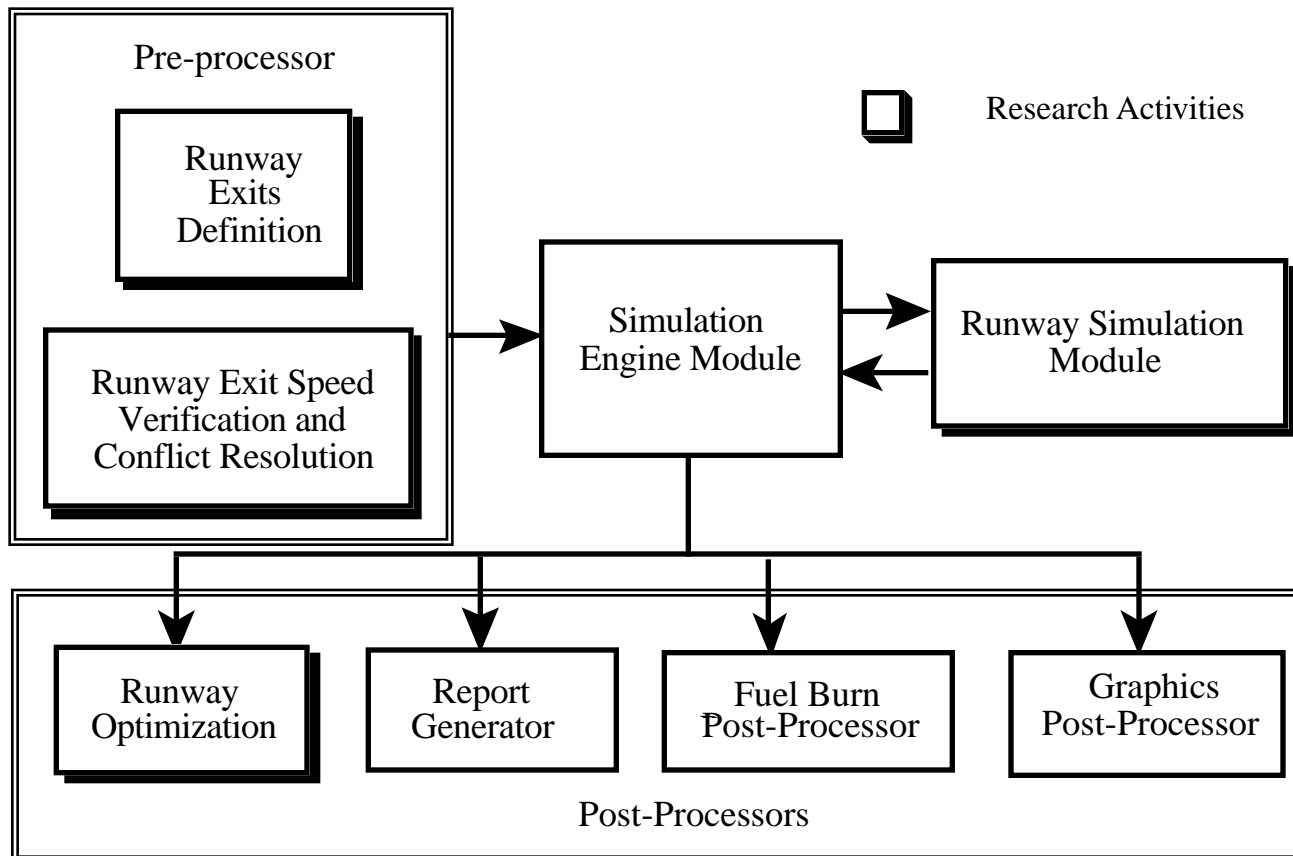
- SIMMOD runway operation model implementation



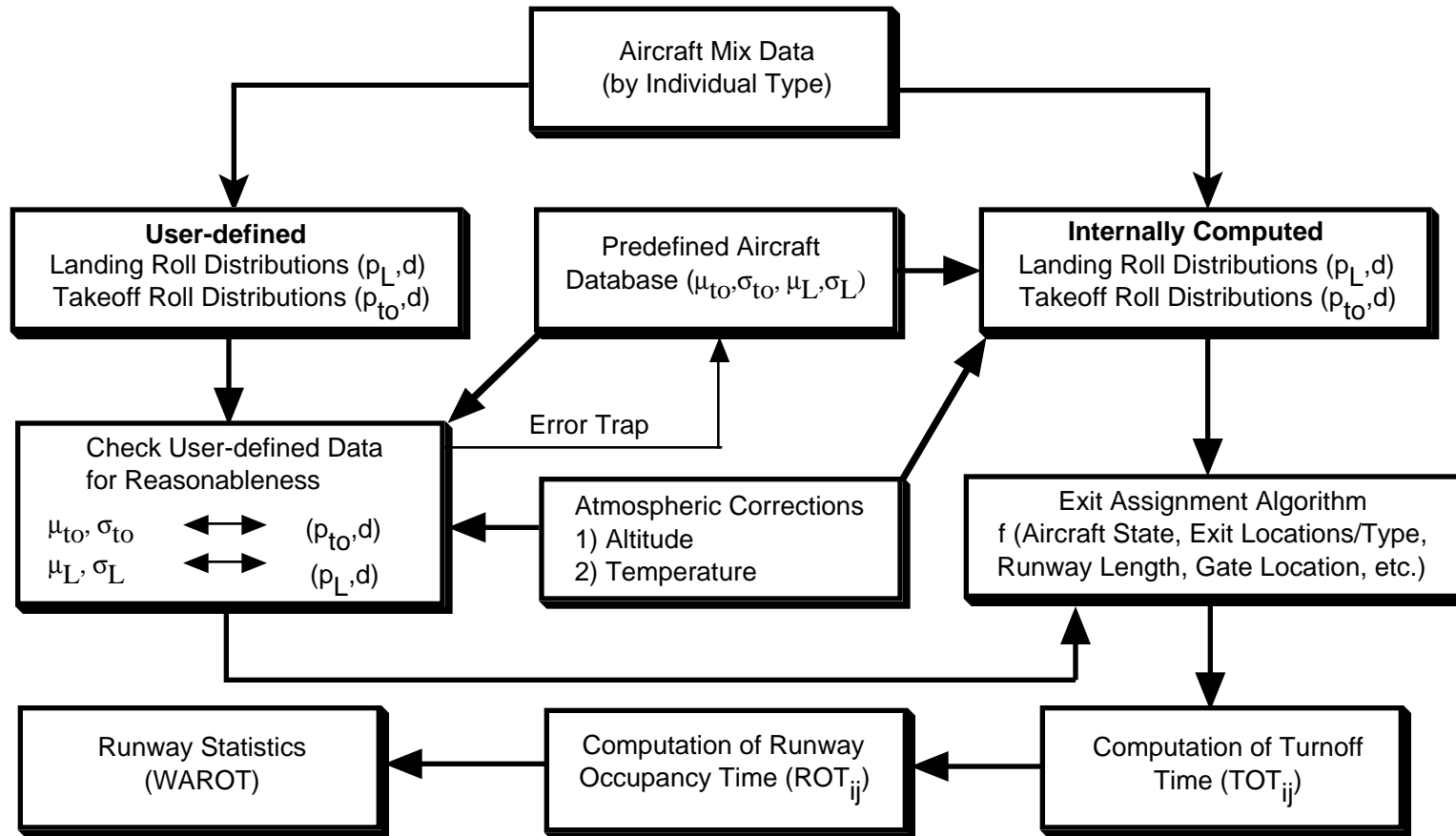


Implementation Issues

- Research Activities to Enhance SIMMOD (for ROT Estimation)



SIMMOD Runway Operations Module



SIMMOD Modules Affected



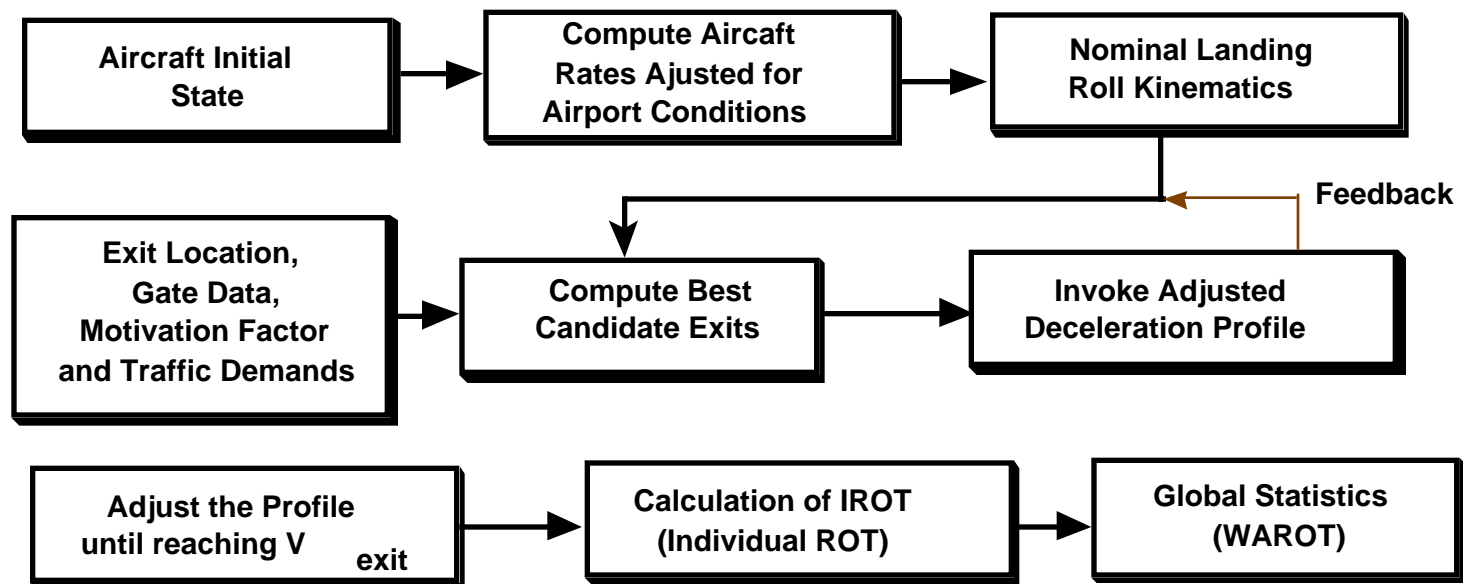
Various modifications were made to the following SIMMOD modules:

- Preamble
- ROUTINE ARR.TAXIWAY.PLANNING
- PROCESS QUE.ARRIVAL
- DISCOVER.RWY.EXIT
- ROLL.TIME

Pilot Decision Making Module



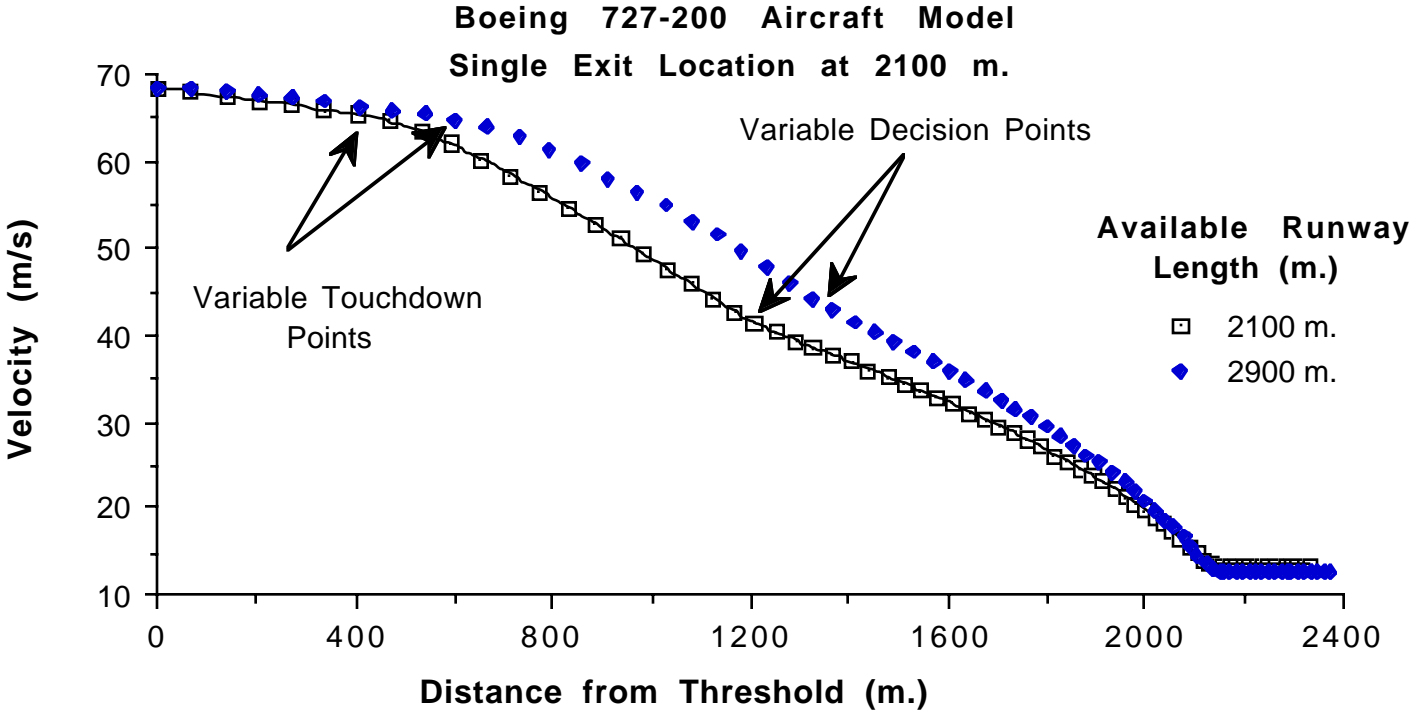
Several exit choice models were developed for SIMMOD considering aircraft performance, airline motivational practices, exit location proximity and ground traffic state conditions



Use of a 2nd Order Feedback Model



A simple pilot-vehicle model adjusts the aircraft kinematics to model the behavior on the runway



Takeoff Roll Time Estimation

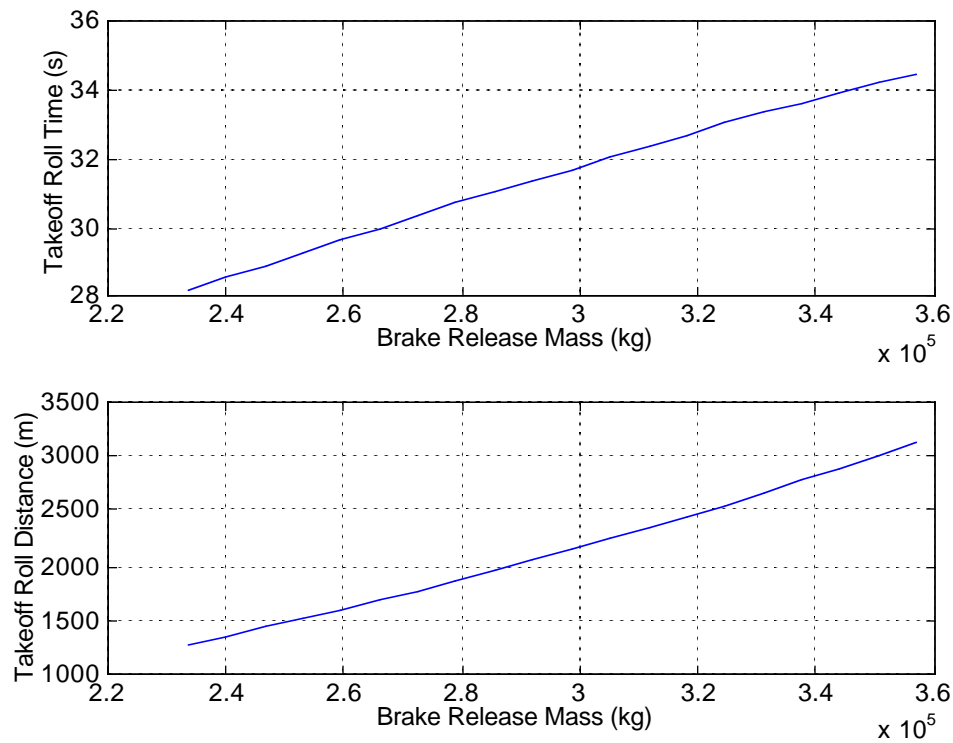


- Use a dynamic takeoff roll model derived from the static SAE 1847 takeoff roll procedure employed in INM
- Use numerical integration of simplified aerodynamic coefficients along the runway
- Procedure is accurate yet simple enough to be introduced into a fast-time simulation model like SIMMOD

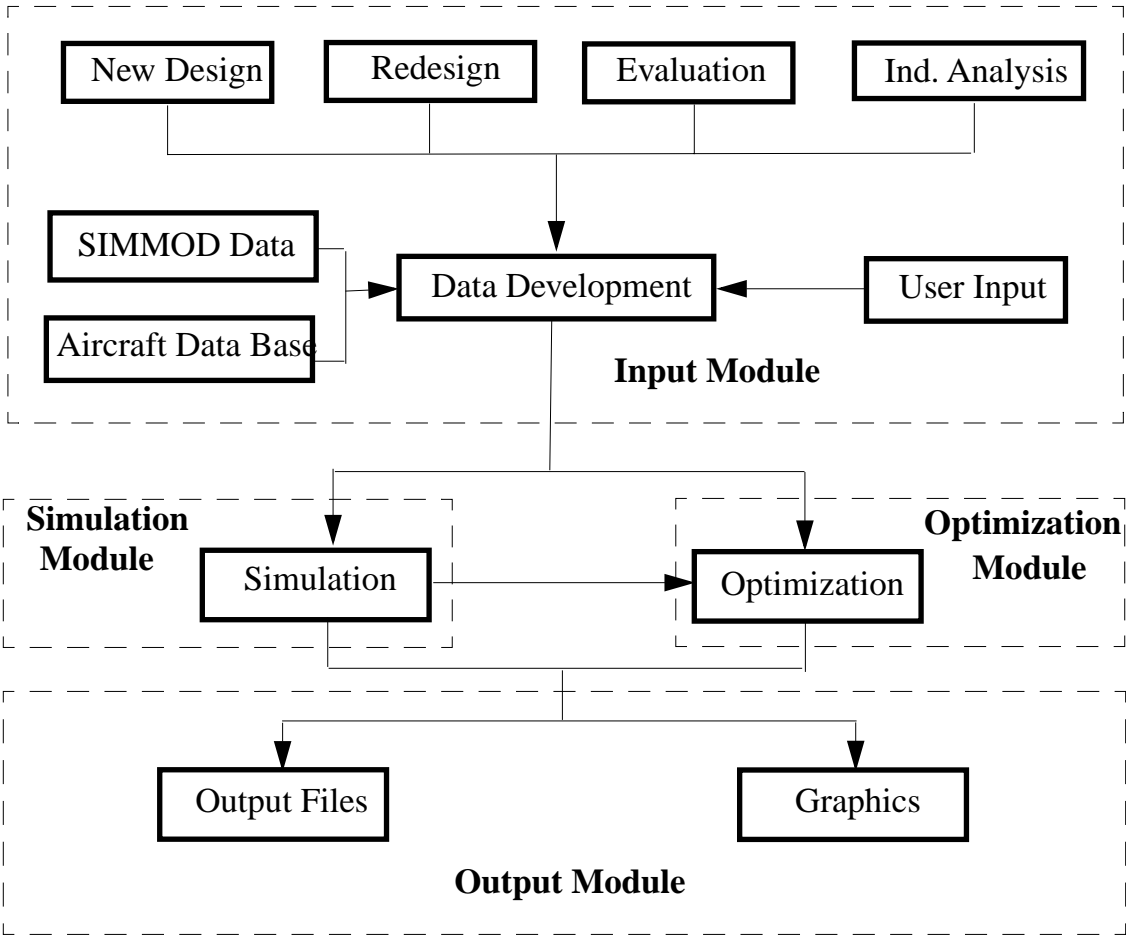
Sample Takeoff Roll Times and Distances



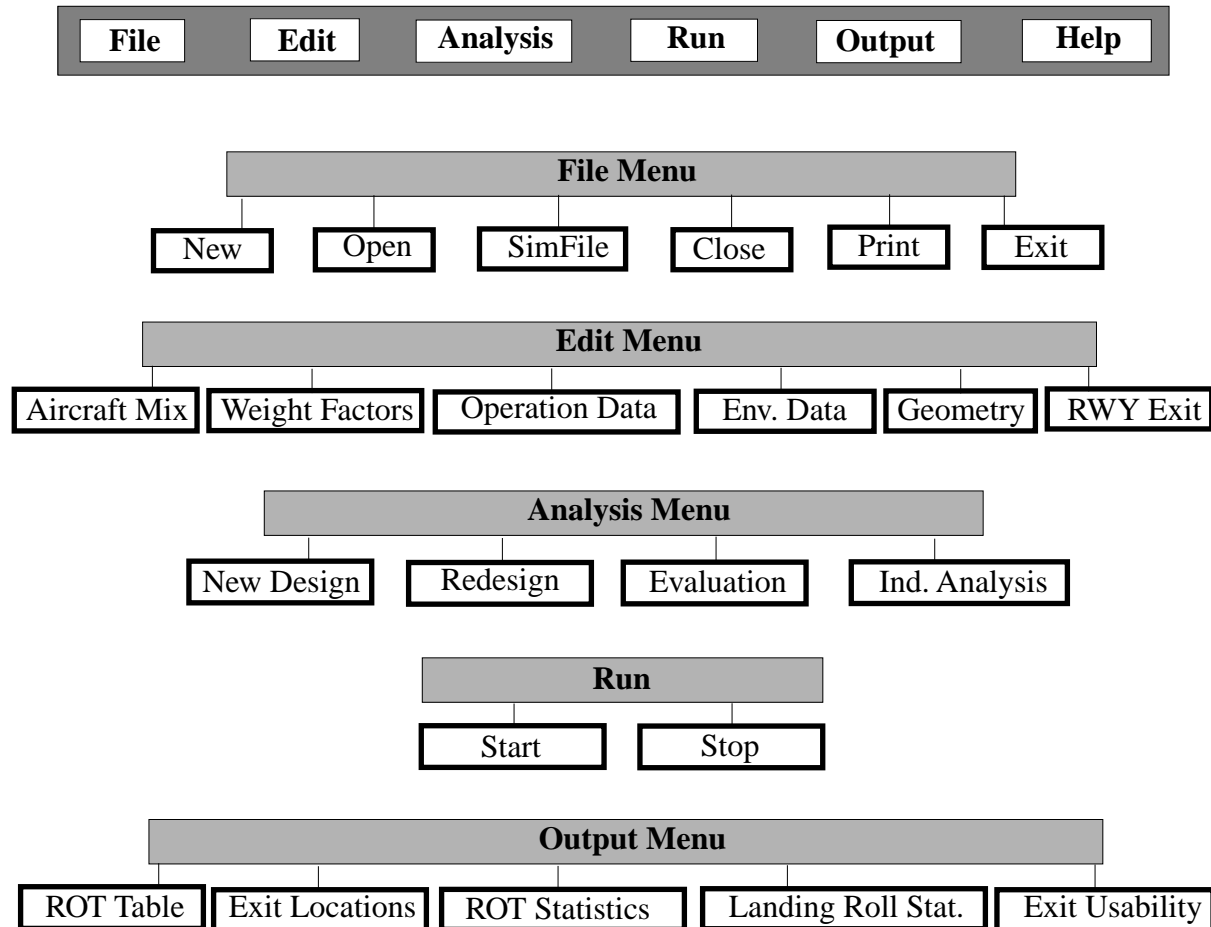
The example illustrates TO roll time and distance for a Boeing 747-200 aircraft



SIMMOD Runway Analysis Module



SIMMOD ROT Pre-Processor Structure

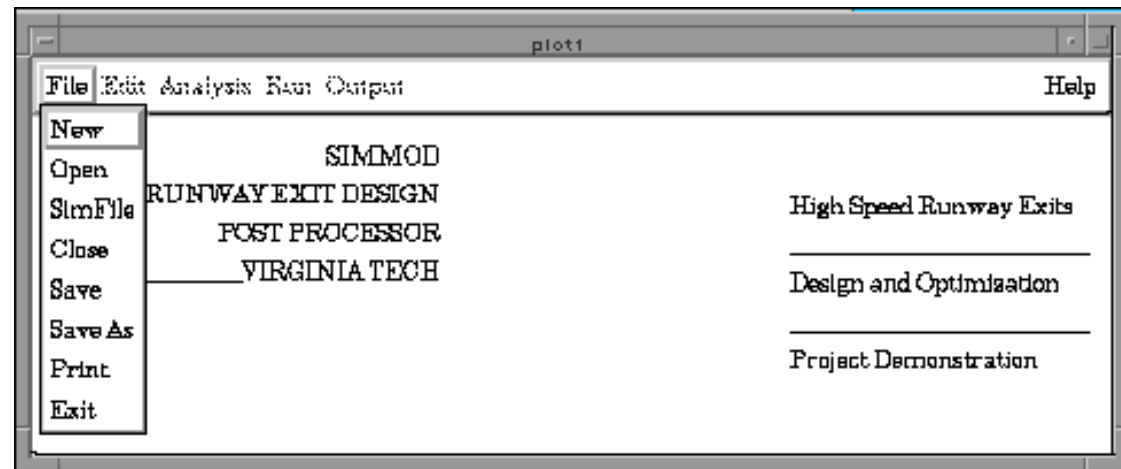




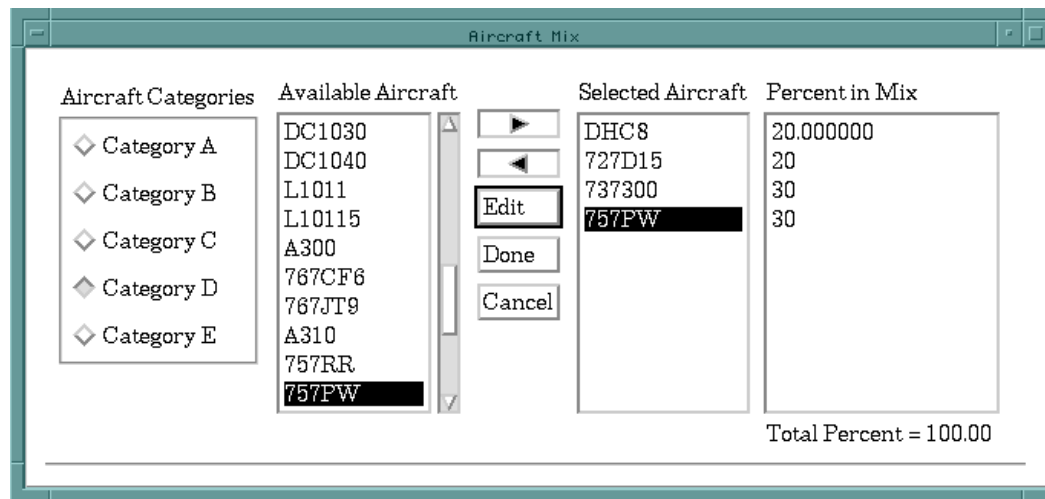
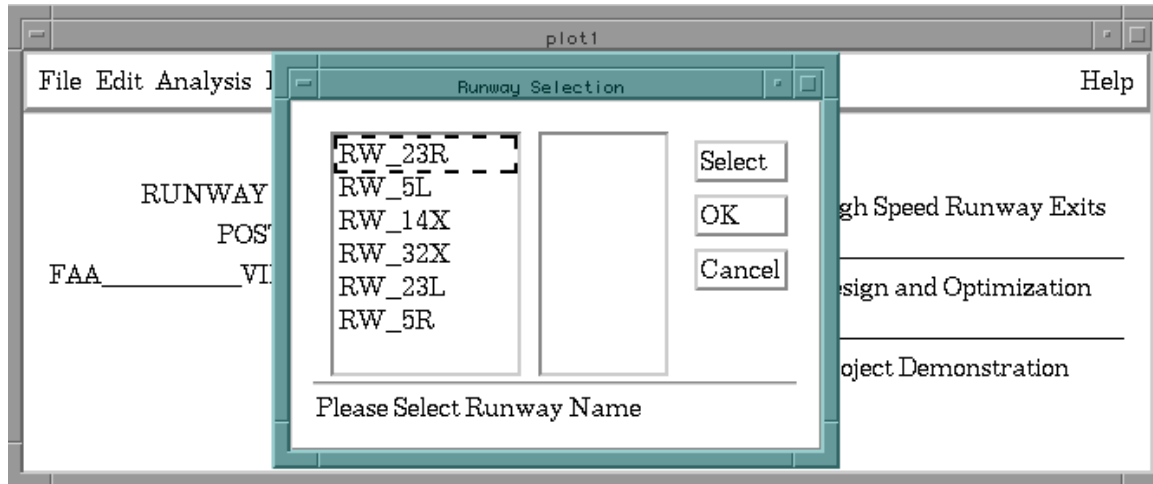
Sample GUI Screens

All module development done in the C language and MOTIF standards (for the GUI) - portability was an issue

Original implementation (IBM AIX 4.5) later ported to Solaris 2.5



Runway and Aircraft Mix Screens



Runway Exit Data Entry Screen



Exit's Data

Existing Exits Data _____ Location _____ Speed _____

Existing Exit No. 1	<input type="checkbox"/> 30 Degree	<input type="checkbox"/> 90 Degree	<input type="checkbox"/> Wide throad	<input type="checkbox"/> Open	<input type="text" value="1000.000"/>	<input type="text" value="14.00000"/>
	<input type="checkbox"/> 45 Degree	<input type="checkbox"/> 30 Redim	<input type="checkbox"/> User defined	<input type="checkbox"/> Close		
Existing Exit No. 2	<input type="checkbox"/> 30 Degree	<input type="checkbox"/> 90 Degree	<input type="checkbox"/> Wide throad	<input type="checkbox"/> Open	<input type="text" value="2000.000"/>	<input type="text" value="10.00000"/>
	<input type="checkbox"/> 45 Degree	<input type="checkbox"/> 30 Redim	<input type="checkbox"/> User defined	<input type="checkbox"/> Close		
New Exit No. 1	<input type="checkbox"/> 30 Degree	<input type="checkbox"/> 90 Degree	<input type="checkbox"/> Wide throad			<input type="text" value="14.00000"/>
	<input type="checkbox"/> 45 Degree	<input type="checkbox"/> 30 Redim	<input type="checkbox"/> User defined			
New Exit No. 2	<input type="checkbox"/> 30 Degree	<input type="checkbox"/> 90 Degree	<input type="checkbox"/> Wide throad			<input type="text" value="14.00000"/>
	<input type="checkbox"/> 45 Degree	<input type="checkbox"/> 30 Redim	<input type="checkbox"/> User defined			

OK

Cancel

ROT Table Output



Report: New Design Case

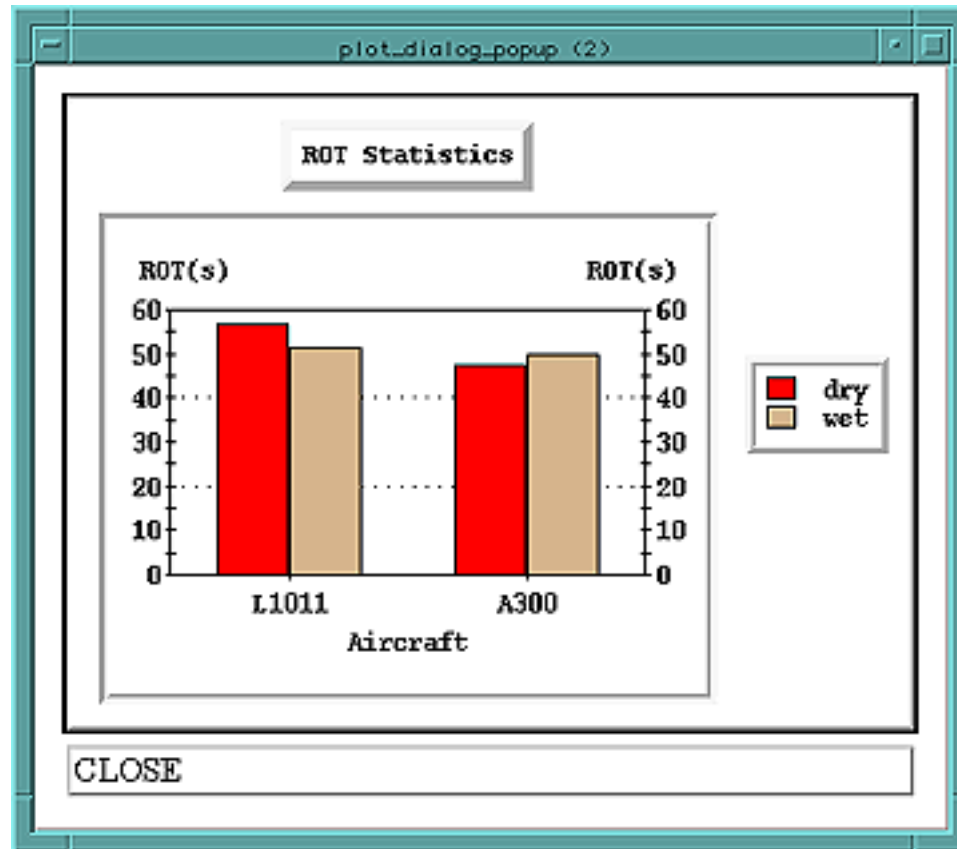
Report: Runway Occupancy Time Table

Exit #	1	2	3
Location(m)	1412.50	1912.50	3000.00
Upper Bound	1325.00	1825.00	3000.00
Lower Bound	1500.00	2000.00	3000.00
Open/Close	Open	Open	Open
Exit Type	30 Degree	30 Degree	90 Degree
DHC8	1	2	3
Dry	53.80	0.00	0.00
%	1.00	0.00	0.00
Wet	54.20	0.00	0.00
%	1.00	0.00	0.00
737300	1	2	3
Dry	48.90	57.80	0.00
%	0.84	0.16	0.00
Wet	49.70	58.80	0.00
%	0.18	0.51	0.00

The weighted average ROT = 54.276489

Ok

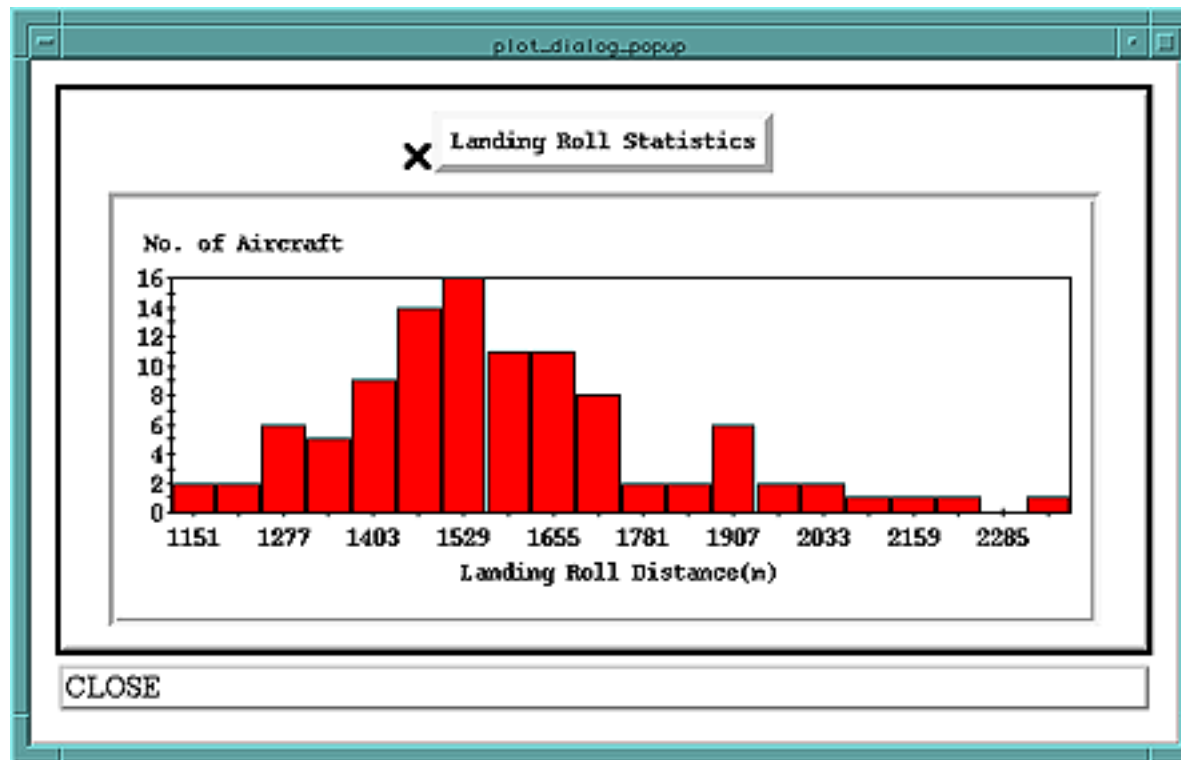
ROT Graphical Output



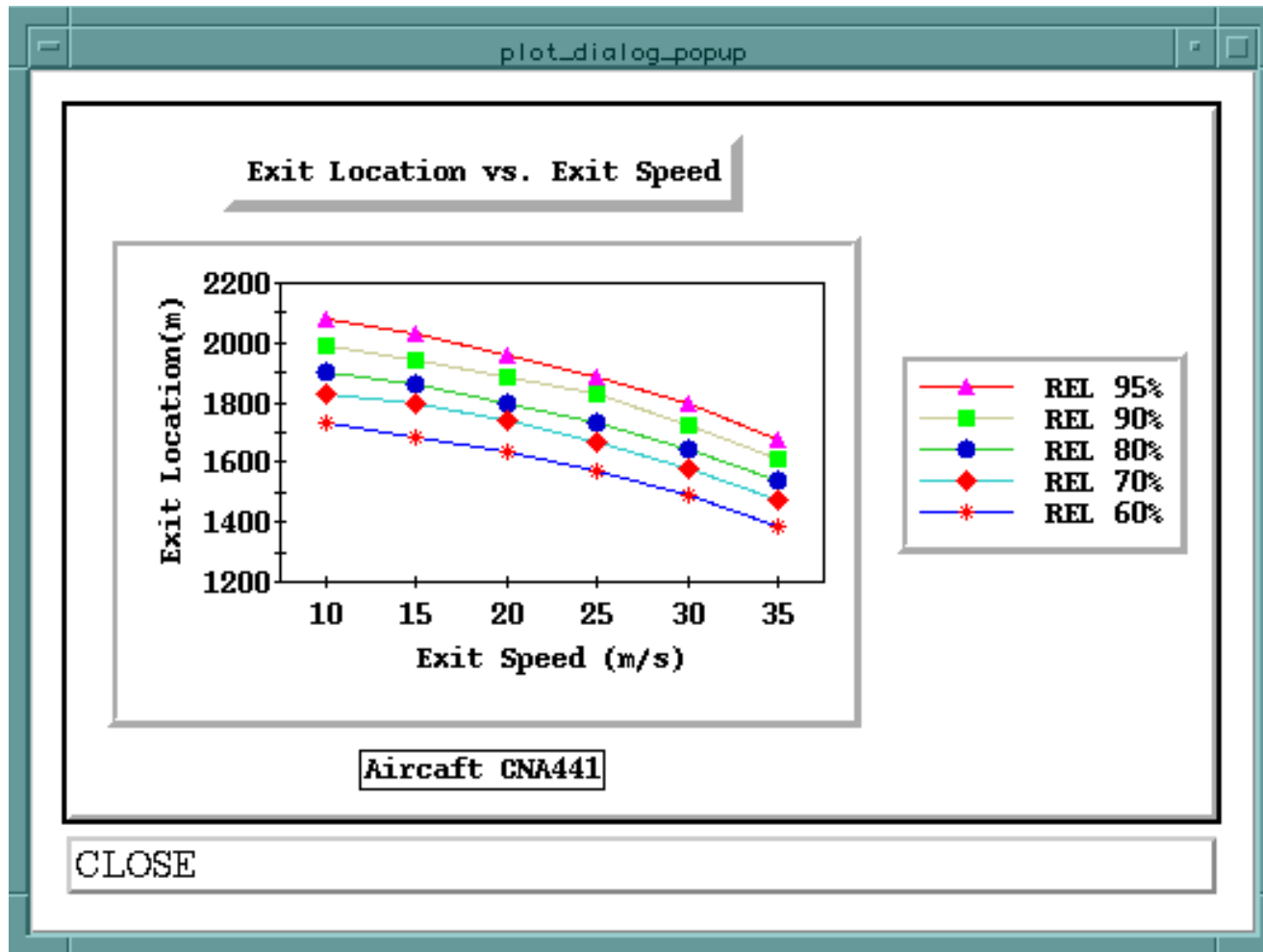
Aircraft Landing Roll Statistics



- Example of a SIMMOD runway operations pre-processor developed at VPI



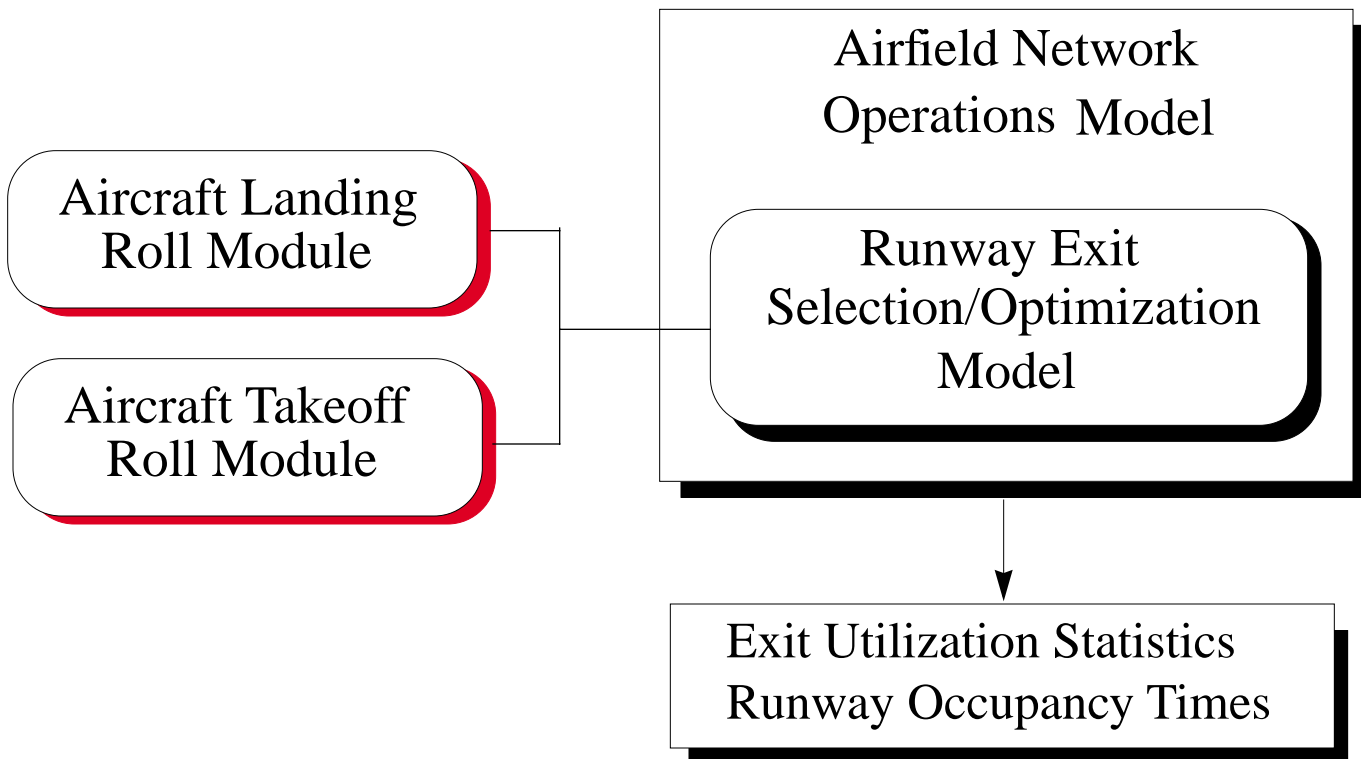
Exit Speed vs. Exit Location Trade-off Plot



Model Extensions (Gate Location)



- An extension of the runway model accounts for pilot motivational behaviors due to gate location constraints



Model Extensions (Gate Location Constraint)



A multiobjective integer programming optimization model has been developed. Minimizing the aircraft runway occupancy time (ROT) and taxiing time (TT) are the two objectives of this model. The model can be described mathematically as follows:

$$\text{Minimize } \sum_{i=1}^n (ROT_{ik} + wfTT_{ik})x_i$$

$$\text{Subject to } \sum_{i=1}^n x_i = 1$$

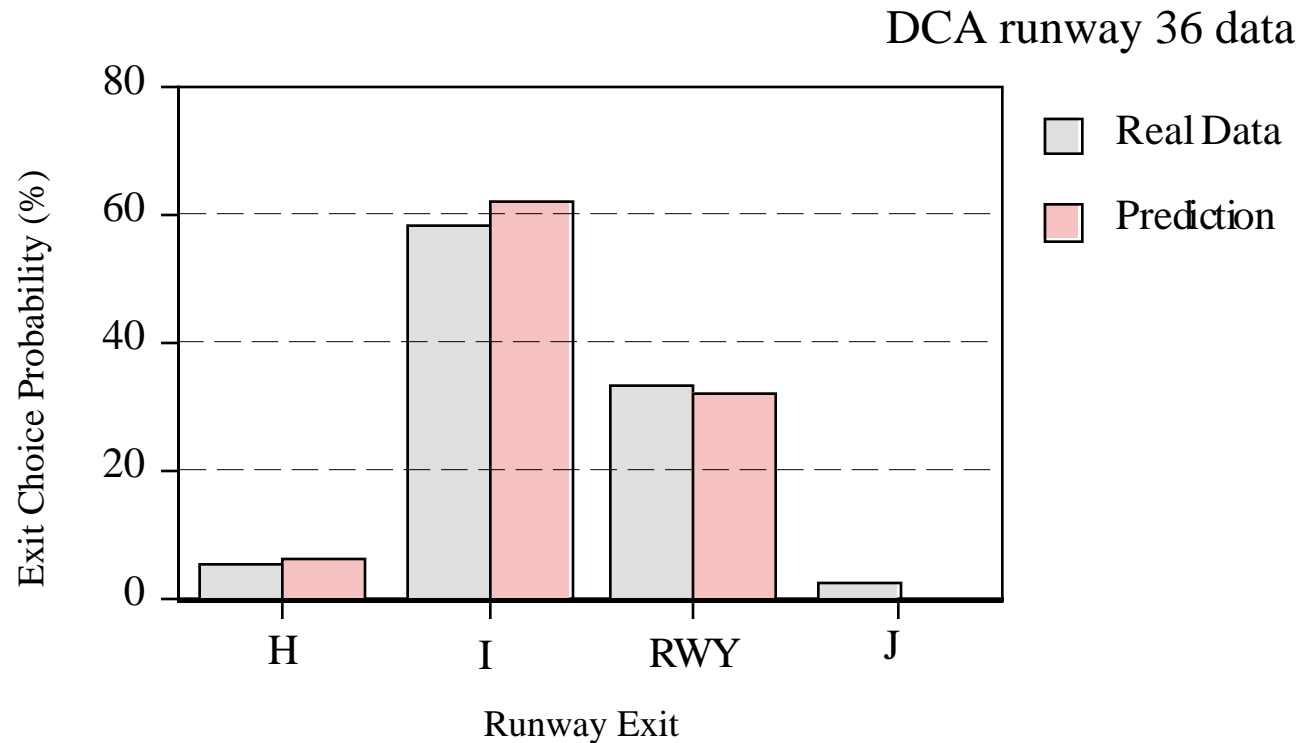
$$l_{ex}(i) \geq S_{air} + S_{fr1} + S_{fr2} + S_{br}$$

$$x_i = 0 \quad \text{or} \quad 1$$

Model Extension Results (Gate Motivation)



- Observed vs. predicted runway exits at DCA



Final Remarks



- The refined SIMMOD model realistically predicts runway service times for all possible runway/taxiway configurations (i.e., aircraft mix and runway/turnoff layouts)
- With these improvements, the new SIMMOD model airport planners could perform sensitivity analyses of various turnoff alternatives based upon delay and fuel consumption evaluations
- Long-term benefits of high-speed turnoffs could be further assessed in terms of landing and takeoff delay analysis
- The new term Dynamic Runway Occupancy Time (DROT) has renewed interest to FAA and NASA